

## NASA SMD Rideshare Office SPRSA September 13-15, 2022

Aly Mendoza-Hill – NASA SMD Rideshare Lead

### NASA Science Mission Directorate Rideshare Office

- The Science Mission Directorate has a rideshare policy and an established SMD Rideshare Office (SRO) to develop standard rideshare processes for the NASA SMD.
- SPD-32 Evolved Expendable Launch Vehicle (EELV) Secondary Payload Adapter (ESPA) Secondary Payloads Rideshare
  - Baseline policy signed Oct. 2018; Policy updated January 2021 (Rev 2); Rev 3 underway... ETA early 2023
    - SMD policy enables rideshare or launch accommodation opportunities using an ESPA-type ring integrated on the launch service procured for an SMD primary payload with identified excess performance.
    - SMD may offer any excess capacity not utilized for SMD investigations to other NASA Mission Directorates (MD), other U.S. Government Agencies, or NASA's International partners in accordance with international agreements for international collaborative efforts relating to science, technology, and exploration goals.
    - \*\* This policy only applies to ESPA-class and independent Cubesat missions on ESPA ports (not Cubesats which are managed through the CubeSat Launch Initiative-CSLI).

## NASA SMD Rideshare Office (SRO)

**Goal:** To provide a single POC for SMD Rideshare-related inquiries for both NASA Center and external partners; to maintain overall knowledge and tracking of Rideshare activities for SMD missions, and to ensure best utilization of excess LV performance to obtain maximum science on SMD missions

#### Located within the Heliophysics Division

- Supports ALL SMD Divisions
- > Aly Mendoza-Hill is the Rideshare Lead for SMD
  - Other key SMD Rideshare Team Members: Alan Zide, David Cheney, Pete Wilczynski, Florence Tan, & Brooke Burns
- Works closely with the Launch Services Program
- Works with NASA Center Rideshare POCs to create unified NASA/SMD Rideshare message; delegates tasks to appropriate Center POCs as required; does not replace Center-level Rideshare work
- Developing a robust rideshare program to utilize excess mass to orbit and enable additional launch opportunities for the science community
  - Standardizing Announcement of Opportunity (AO) language and reviewing each AO for consistency
  - Developing key documents: SMD RS101, SMD RUG & DNH requirements, & the SMD RS Implementation Plan
  - Performing top-level payload compatibility analyses of rideshare missions to identify potential impacts to the primary payload or the success of the secondaries
  - Maintaining a list of SMD launch opportunities and tracking potential external launch opportunities
    - External information is made available on the Small Spacecraft Systems Virtual Institute (S3VI) website (NASA Launch Portal - <u>https://www.nasa.gov/smallsat-institute</u>) <sup>3</sup>





## **Rideshare Launch Opportunity Primary POC's**



# NASA HQ Rideshare

To implement NASA's Mission, NASA Headquarters is organized into five principal organizations called Mission Directorates:

• Aeronautics: Pioneers and proves new flight technologies that improve our ability to explore and which have practical applications on Earth.

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Sam Johnson\*

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Chris Baker

- Exploration Systems Development: Development programs for deep space exploration, including Artemis missions.
- Space Operations: Space operations in low-Earth orbit.
- Science: Explores the Earth, moon, Mars, and beyond; charts the best route of discovery; and reaps the benefits of Earth and space exploration for society.
- Space Technology: A calalyst for the creation of technologies and innovation needed to maintain NASA leadership in space while also benefiting America's economy.
- Mission Support: Oversees the management of the institutional functional areas that support the Agency mission.

### Access to Space Process for NASA SMD Launches









	SMD Missions: <u>Potential</u> Rideshare Opportunities Note: Dates are estimates and subject to change, excess launch capacity on most flights still TBD						
	Mission	Org	LV Class	Trajectory	LRD	Rideshare Adapter	RPLs
	GOES-U	JASD	Falcon-Hvy	GTO (8000x35000) ~10 deg	04/30/24	TBD	TBD - excess performance being evaluated
	COSI	APD	TBD	LEO 550km 5deg	September 2025	TBD	TBD
	Sentinel 6B	ESD	Intermed/Larg e (tbd)	LEO (1336km; 66deg (tbd)	Nov 2025	TBD	TBD
	NEO Surveyor	PSD/PDCO	Intermed/Larg e (tbd)	L1 (C3 tbd)	May 2026	TBD	TBD
	JPSS-3	JASD	Intermed/Larg e (tbd)	LEO SSO 13:30 MLTAN	2025 or 2027	TBD	TBD
	INCUS	ESD	Small/Interme diate	500km 28.5 deg	Oct 2026	TBD	TBD
	MUSE	HPD	Intermed/Larg e (tbd)	SSO 620km 97.9 deg 6am MLTAN	June 2027	TBD	TBD
	GDC	HPD	Intermed/Larg e (tbd)	400km, 81-82 deg	May 2028	TBD	TBD
	2021 MIDEX AO	APD	Intermed	TBD	NLT Dec 2028	TBD	TBD
	HelioSwarm	HPD	Intermed	C3 = 2.75 km2/s2	2028/2029	TBD	TBD
	DAVINCI+	PSD	Intermed/Larg e (tbd)	Venus (C3=18)	06/01/29	TBD	TBD
	VERITAS	PSD	Intermed/Larg e (tbd)	Venus (C3=17)	tbd (2028/2030)	TBD	TBD
Public Mission List location: Small Spacecraft Virtual Institute – NASA Launch Portal <u>https://www.nasa.gov/smallsat-institute</u> 122 For NASA Internal ONLY							

# NASA SMD RS & SmallSat – Top 5 technical challenges



- Early coupled loads analyses
  - 50-75Hz fundamental frequencies
  - Random Vibe High frequencies from SX
- DSN coverage post SC separations L1 vs LEO
- Contamination Control every SC has a different set of requirements. (Instrument GN2 flowrate & humidity, magnetic cleanliness, silicone sensitivity, hydrocarbon, surface cleanliness, etc)
- Rideshare payload power-on for extended coast phases (thermal)
- On-board propulsion for NASA 'unique' orbits
  - vs relying on LV delta-v
  - Reliability and TRL of propulsion systems for deep space

## **Rideshare Challenges & Lessons Learned:**

National Aeronautics and Space Administration



Challenges & Lessons Learned:

- •Utilize excess performance on NASA SMD Primary launches => RS Policy
- NASA Science-unique orbits => VADR
   Promote expansion of commercial launch options for SmallSats with higher risk tolerance
- Partnering with government and commercial providers for increased opportunities
  Compatibility and DNH for reducing risks to all
- •Compatibility and DNH for reducing risks to a payloads
- •Mission flexibility for increasing launch opportunities and reducing costs

#### NASA SmallSat Launch Options

 SMD Rideshare opportunities
 LSP Commercial VADR launch contract for SmallSats for Dedicated Launch & Rideshare
 VADR = Venture-Class Acquisition for Dedicated and Rideshare Launch Service contract Current Opportunities & S3VI Resources:

- NASA Launch Portal. (<u>https://www.nasa.gov/smallsat-institute/launchportal</u>)
- Upcoming SmallSat Mission solicitations. (<u>https://www.nasa.gov/smallsat-institute/nasa-smallsat-opportunities</u>)
- NASA SMD Rideshare Guide (<u>https://www.nasa.gov/sites/default/files/atoms/files/s</u> md spa rug with dnh generic 2021dec15.pdf)

For questions about NASA launch opportunities or other potential rideshare opportunities, email:

hq-smd-rideshare@mail.nasa.gov

## Backup

- NEW NASA LSP Launch Service Contract for higher risk tolerant missions VADR
  - Venture-class Acquisition for Dedicated & Rideshare
- Example SmallSat Rideshares Pioneers & TRACERS
- GeoCarb Info (Large SC to GTO/GEO)

### VADR - Venture-class Acquisition for Dedicated & Rideshare

- NEW NASA LSP Launch Service Contract for higher risk tolerant missions
  - IDIQ to procure FAA licensed commercial launch services for Class D/high risk tolerant payload and streamlined CubeSat launch services
    - Less technical oversight (insight & approval) over the launch service than on nominal NLS-II launch services for less risk-tolerant SMD primary missions... i.e. limited LSP mission assurance
  - Launch providers bid either a dedicated or rideshare solution based on the particular mission's requirements.
    - Rideshare solutions can either be "Primary RS" or "Traditional RS"
      - Primary RS controls the orbit and the launch date; Traditional does not
  - Multiple Providers/Launch Vehicles
  - First flight not required to bid

# VADR Contracts Awarded Initially

ABL Space Systems	Relativity Space, Inc.	
Astra Space, Inc.	Rocket Lab USA, Inc.	
Blue Origin Florida, LLC	Spaceflight, Inc.	
L2 Solutions, LLC	Space Exploration Technologies Corp.	
Northrop Grumman System Corp.	United Launch Services, LLC	
Phantom Space Corp.	Virgin Orbit, LLC	

## Example SmallSat Rideshares - 2020 APD Pioneers

PANDORA	ESPA Mass: 115 kg Volume: 1.26m x 0.98m x 0.90m	Orbit: LEO SSO Altitude: 500-600km (preferred) / 450-600km (acceptable) Inclination: SSO (97.4-97.79 preferred; 97.03-98 acceptable) LTAN: 6pm (preferred) / 5am-7am and 5pm-7pm (acceptable) Launch date: October 2024	<ul> <li>ISO 8</li> <li>GN2 purge only thru mate/encapsulation</li> <li>No propulsion</li> </ul>
ASPERA	ESPA Mass: 59.2 kg Volume: 0.55m x 0.44m x 0.42m	Orbit: LEO SSO Altitude: 700-900km (preferred) / 530-850km (acceptable) Propulsion system will be used for orbit insertion and de-orbit. Inclination: 98.2-99.0deg SSO(preferred) / Any SSO (acceptable) LTAN: 6am/6pm dawn/dusk (preferred) /Any SSO (acceptable) Launch date: May 2025	<ul> <li>ISO 7</li> <li>GN2 Purge thru T-0: Grade 5 GN2 @ 0.02scfm</li> <li>Indium propulsion. Solid state metal and inert at room temp.</li> </ul>
StarBurst	ESPA Grande Mass: 299.7 kg Volume: 0.90m x 0.80m x 0.92m	Orbit: LEO Altitude: 363-536km (preferred) / 300-600km (acceptable) Inclination: 0deg (preferred) / 0-60 degrees (acceptable) Launch date: November 2024	<ul> <li>IEST-STD-CC1246E: Visibly clean; highly sensitive + UV (VCHS+UBV) level of cleanliness.</li> <li>No propulsion</li> </ul>

#### SMD Rideshare Office – Class D Launch Options Assessment - TRACERS

**TRACERS Mission Information** 

Helio SMEX Date of Selection: Nov 5, 2021

Two(2) Spinning SmallSats flying in formation Mass: 131-200 kg each Volume: .948m x 1.382m diam

Orbit: Polar SSO Altitude: 500km circular Inclination: 97.5 MLTDN 10:30 (preferred)... 02:30 (secondary)... can do 06:00 with additional TRACERS

#### **Mission Uniques:**

- Powered-on at launch and during ascent
- GN2 Purge
- Spin separation & thermal roll requirements
- Free molecular Heating requirement
- Contamination monitoring inside the fairing
- Spare EGSE and umbilical connectors
- Hydrazine Fueling
- Dedicated PPF
- MLTDN preference

**OGA Rideshare Options** 

**?** USSF : nothing yet



#### **Mission Classification & Timelines**

7120.5 Mission ~\$115 PI Cost Cap (not including NASA provided launch)

PDR: Oct 2021 KDP-C: March 31 CDR: July 2022 Launch date: NET July 2024



# GeoCarb – late 2024

#### **Previous - Hosted Instrument Option**

Item	Requirements/Parameters		
Magg	Current Best Estimate (CBE): 155 kg		
IVIASS	Maximum Expected Value (MEV): 179.5 kg		
Current Dimensions	70 in (TBD) x 50.2 in x 44.3 in (door deployed)		
Bus Voltage	100 Vdc		
Orb Average Power	CBE: 393/402W		
(Nominal/Eclipse)	MEV: 518/565W		
Peak Power	CBE: 600W		
Orbit Average Science Data Rate	10 Mbps (fixed rate; redundant LVDS channels)		
Command/Telemetry	MIL-STD-1553B		
GEO Longitude Range	Desired: 85°W +/-10°		
	Allowable: 65°W to 105°W		
Baseline Mission	3 years + 60 days commissioning		
Clear FOV	+/- 10.18 degrees E-W by +/- 9.42 degrees N-S		

Table 1. Current GeoCarb Mission Hosting Reauirements/Parameters

#### **NOW - Full Spacecraft Rideshare Opportunity**

1000-1200kg Spacecraft Bus ("Caketopper" or "Side-by-Side) GTO launch

## NASA HQ Science Mission Directorate Rideshare Office (SRO)

For SMD Rideshare inquiries, contact us at: <u>HQ-SMD-Rideshare@mail.nasa.gov</u>

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